

Micromechanical analysis on transverse compressive behavior and damage mechanism of CF/Al composites considering realistic fiber-arrangement

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April 28, 2020

Abstract

A micromechanical model based on the realistic microstructure of carbon fiber reinforced aluminum (CF/Al) composites was developed for the first time. The transverse compressive behaviors, with particular emphasis on damage mechanism of the composites were investigated by numerical simulation and experiment. The results showed that the micromechanical model considering the realistic fiber arrangement predicts the mechanical properties more accurately than that based on an idealized fiber arrangement, and the calculated stress-strain curves agrees well with the experimental ones. The interfacial damage accumulates with compressive strain increasing, and induces the local interface failure successively. The progression and interaction of interface failure and matrix damage dominates the transverse compression process, and leads to the initiation of fiber failure in the ultimate stage, resulting in a fracture surface with the characteristic of interfacial debonding and fiber rupture. Moreover, parametric analysis based on the micromechanics model was carried out to evaluate the influences of interfacial properties and fiber volume fraction on the transverse compressive behavior of the composites.

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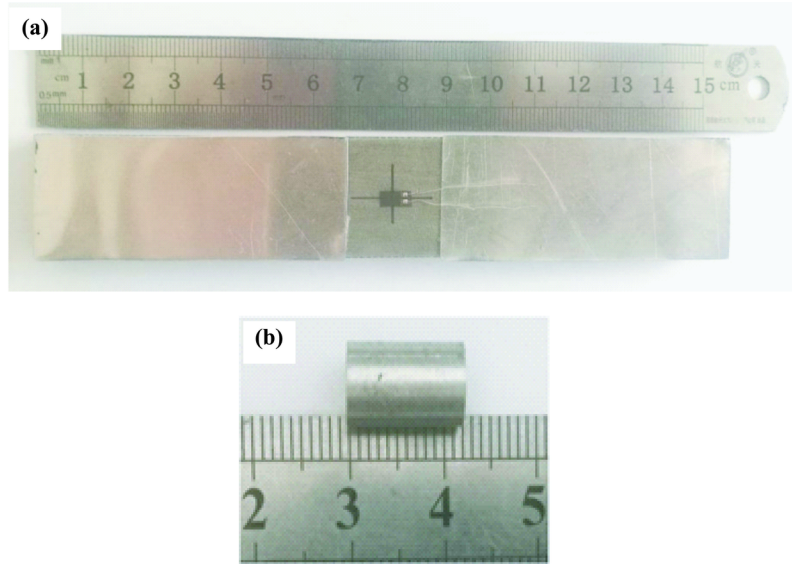


Fig.1 Sample of the CF/Al composite (a) and the ZL301 alloy (b) for uniaxial compression testing

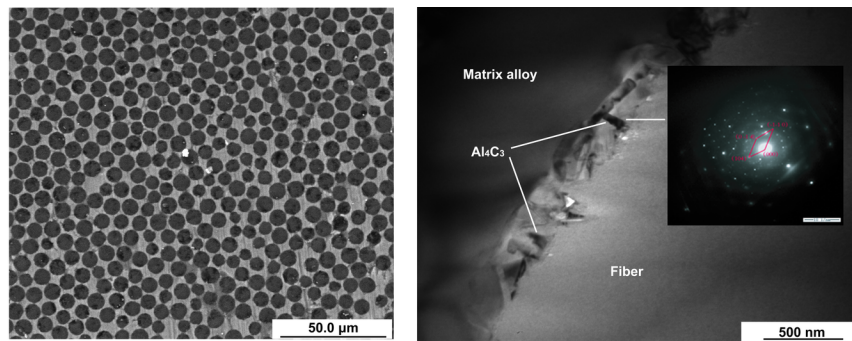


Fig.2 Microstructure of the CF/Al composite (SEM) (a) and the matrix/fiber interface (TEM) in the composite (b). The inset in Figure 2b is the SAED pattern of the interfacial product

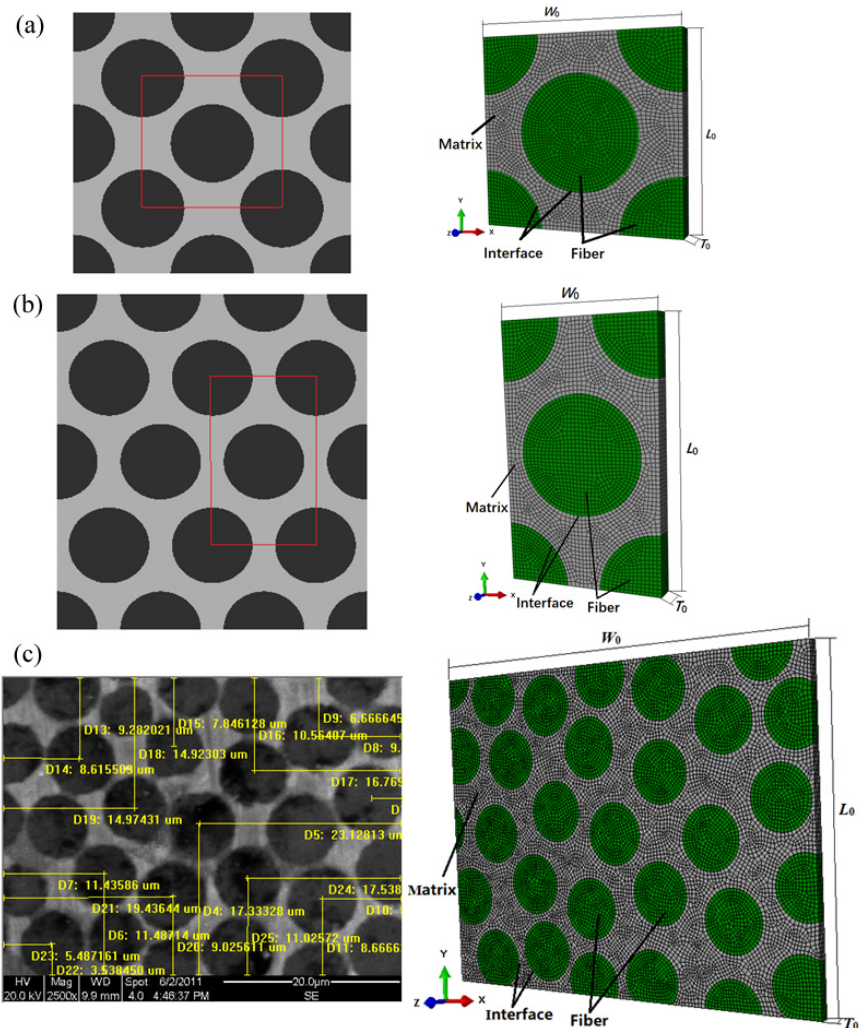


Fig.3 RVE models with the idealized and realistic fiber array geometries
(a) S-RVE model, (b) H-RVE model, and (c) R-RVE model

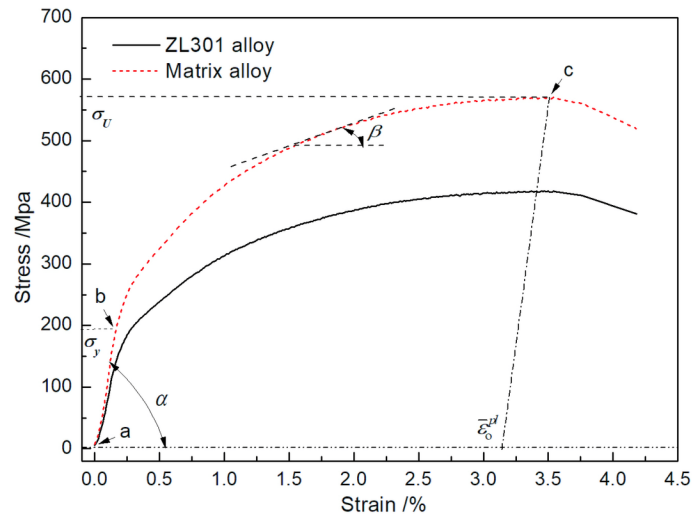


Fig.4 Experimental stress-strain curve of the ZL301 alloy and in-situ compressive curve of the matrix alloy

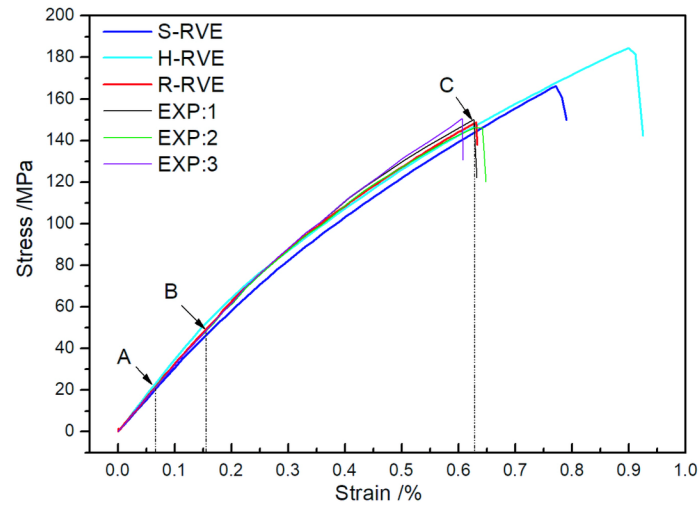


Fig.5 Comparison of transverse compression stress-strain curves for the CF/Al composites: Testing curves and predicted curves by the different RVE models

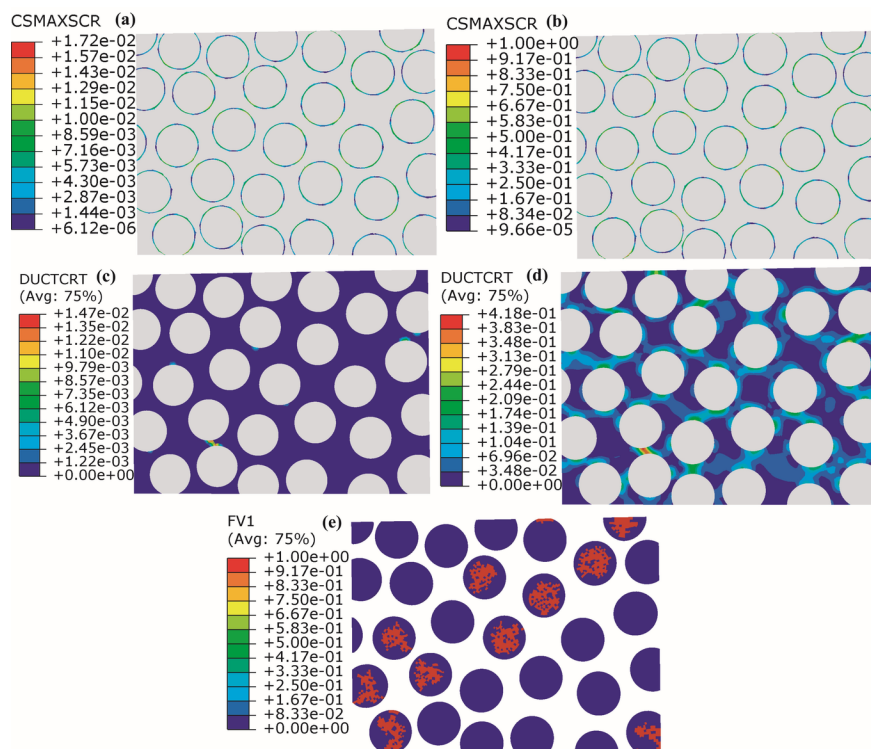


Fig.6 Microscopic damage and failure behaviors of the constituents during transverse compression process: (a)Interfacial damage ($\epsilon=0.001\%$), (b)Interfacial failure ($\epsilon=0.066\%$), (c)Matrix damage ($\epsilon=0.152\%$), (d)Matrix damage ($\epsilon=0.630\%$), (e)Fiber failure ($\epsilon=0.630\%$)

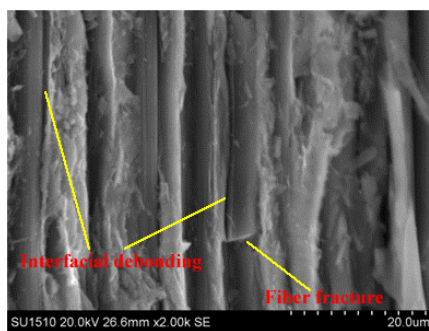


Fig.7 Fracture morphology of the composite under transverse compression condition (SEM)

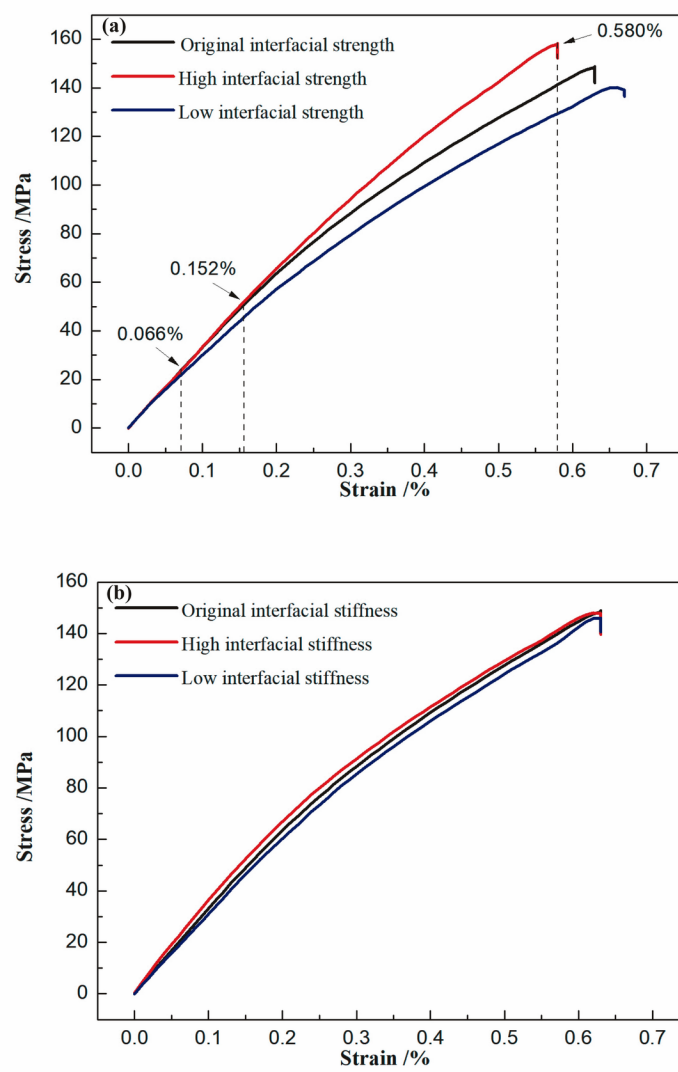


Fig.8 Effect of interfacial strength (a) and interfacial stiffness (b) on the transverse compressive stress-strain behaviors

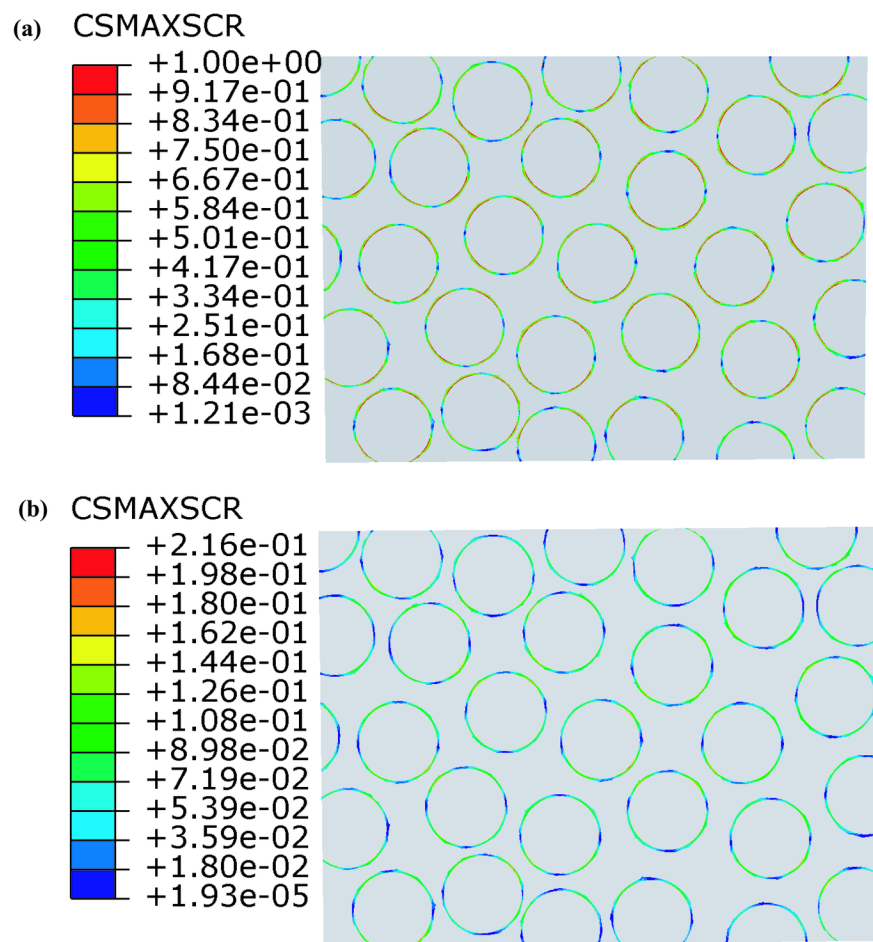


Fig.9 Interfacial damage state at $\epsilon=0.066\%$: (a) the low interfacial strength, (b) the high interfacial strength

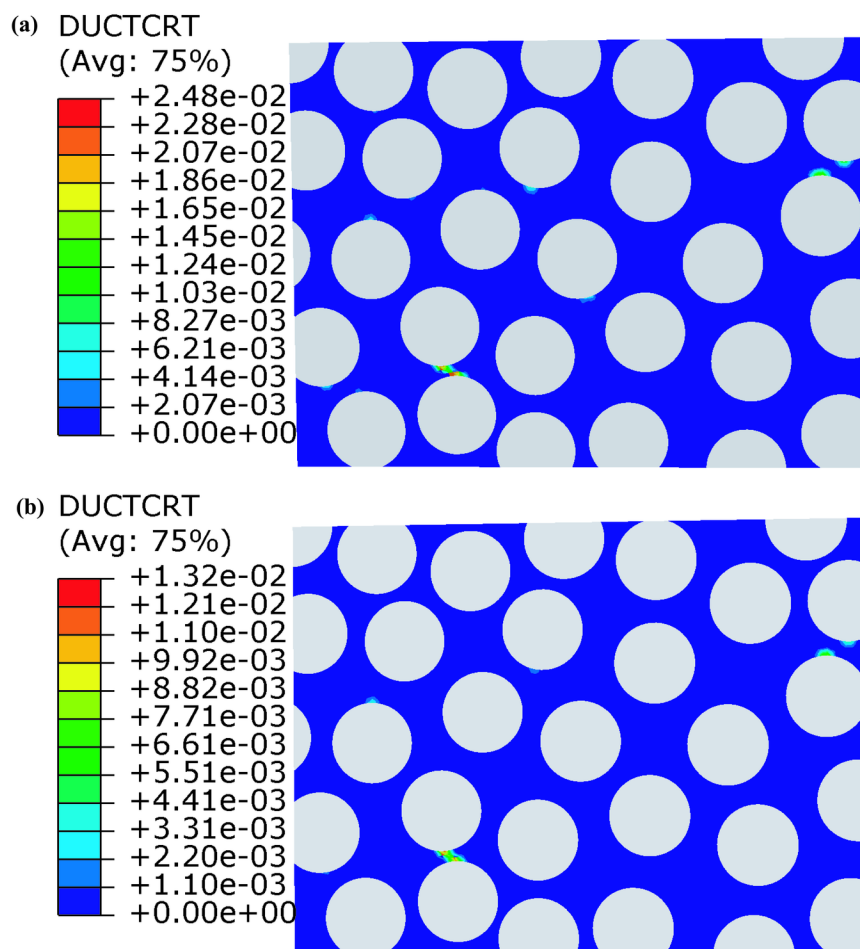


Fig.10 Damage state of the matrix alloy at $\epsilon=0.152\%$: (a) the low interfacial strength, (b) the high interfacial strength

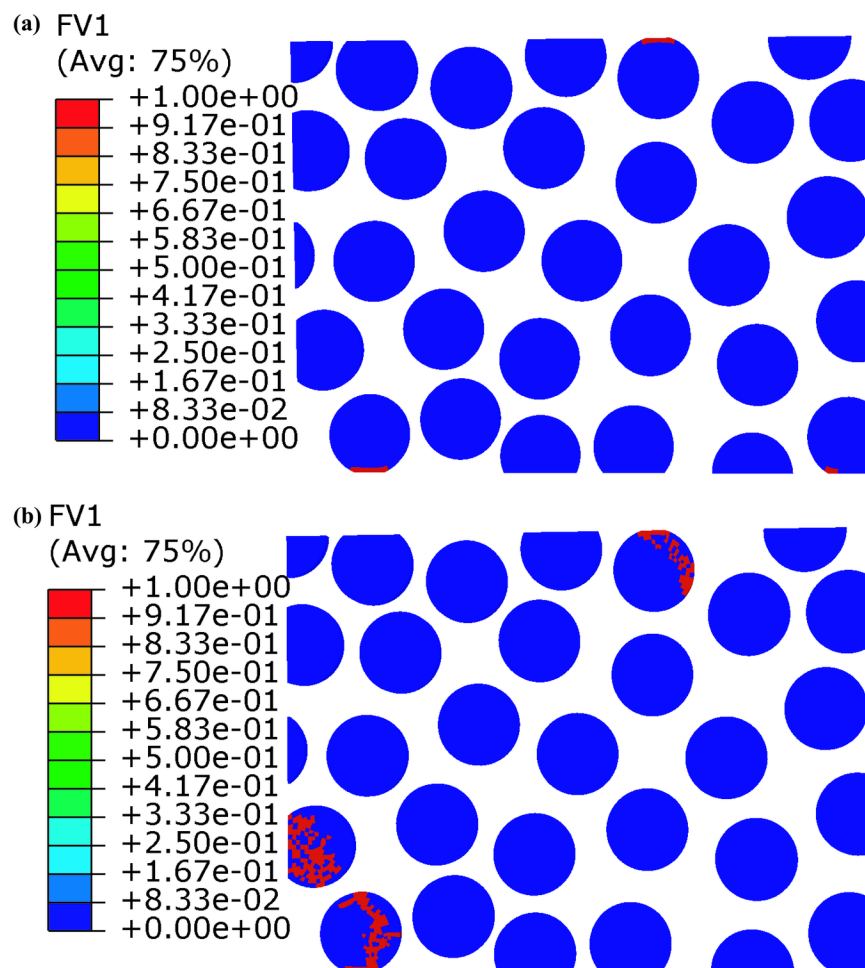


Fig.11 Failure state of the fibers at $\epsilon=0.580\%$: (a) the low interfacial strength, (b) the high interfacial strength

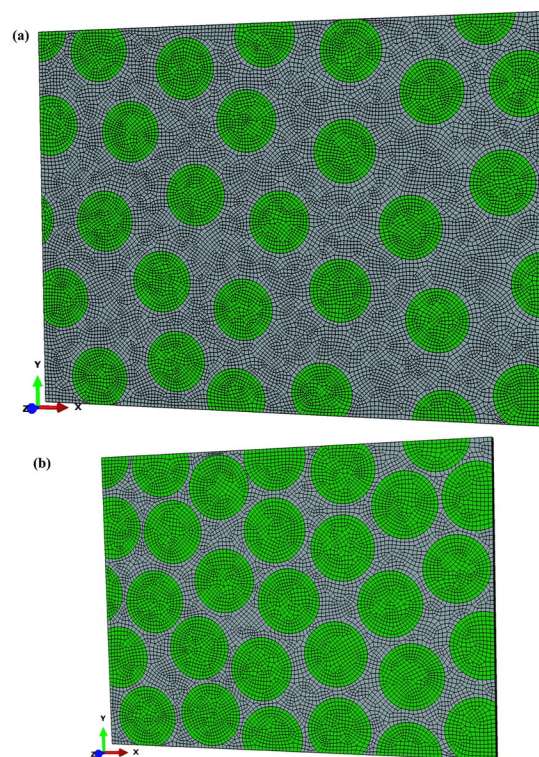


Fig.12 RVE models with fiber volume fraction: (a) $V_f = 40\%$ and (b) $V_f = 70\%$

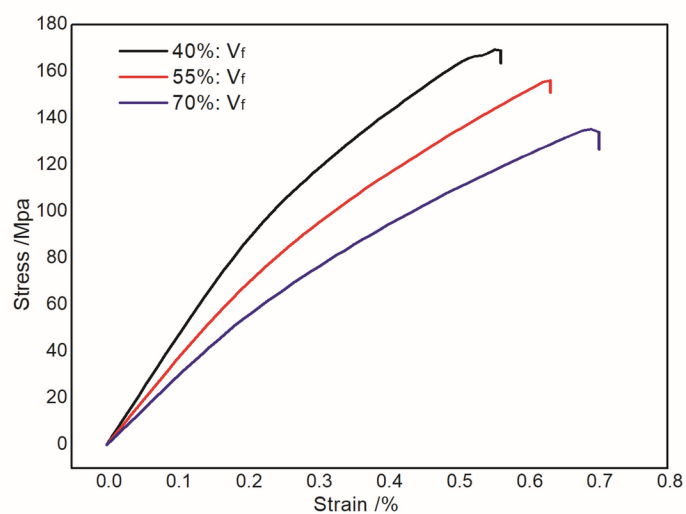


Fig.13 Transverse compressive curves at the different fiber volume fraction conditions

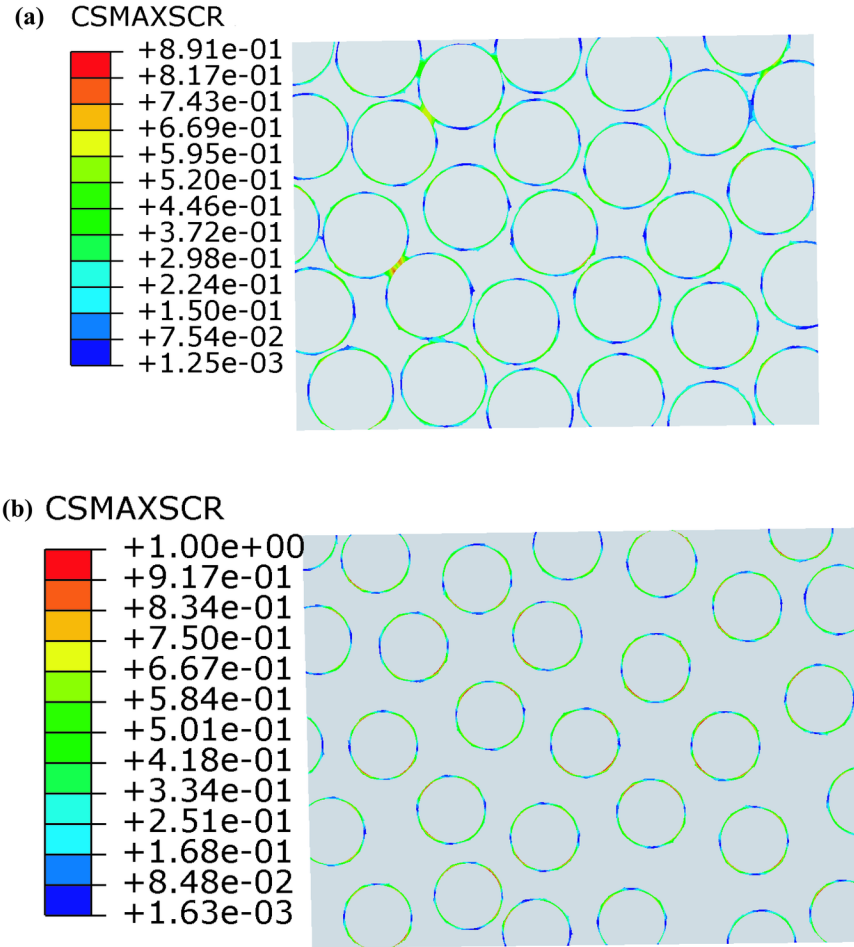


Fig.14 Interfacial damage state ($\epsilon=0.066\%$) at the $V_f=70\%$ (a) and the $V_f=40\%$ (b) conditions

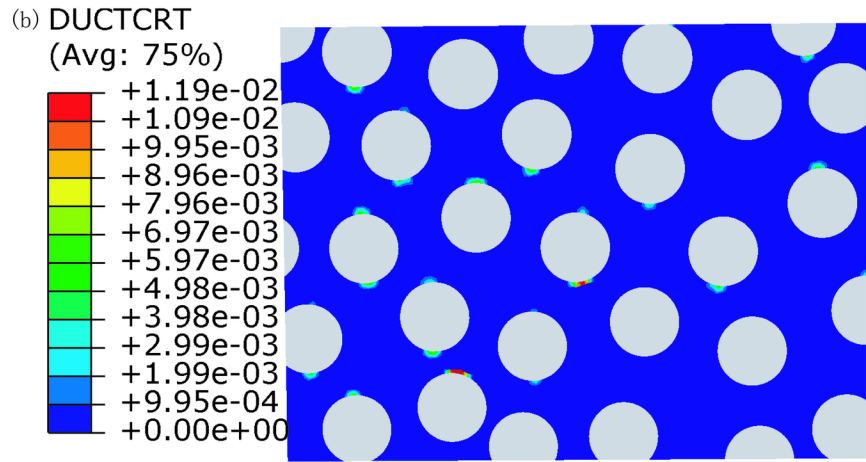
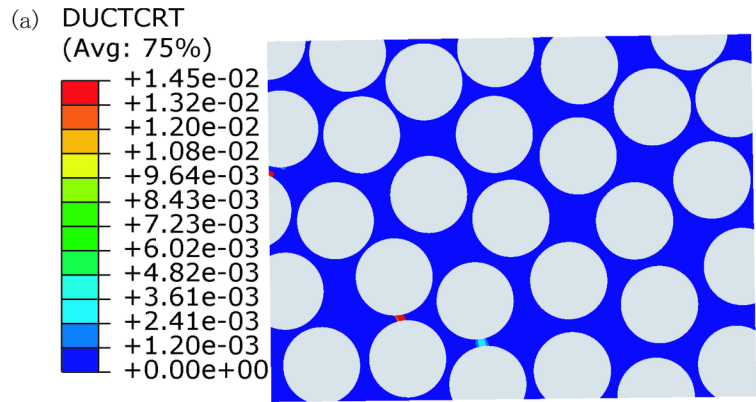


Fig.15 Matrix damage state ($\epsilon=0.152\%$) at the $V_f=70\%$ (a) and the $V_f=40\%$ (b) conditions

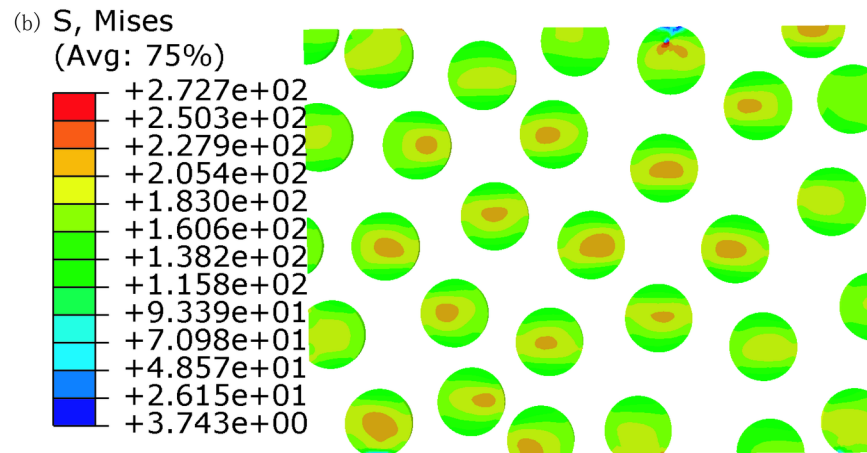
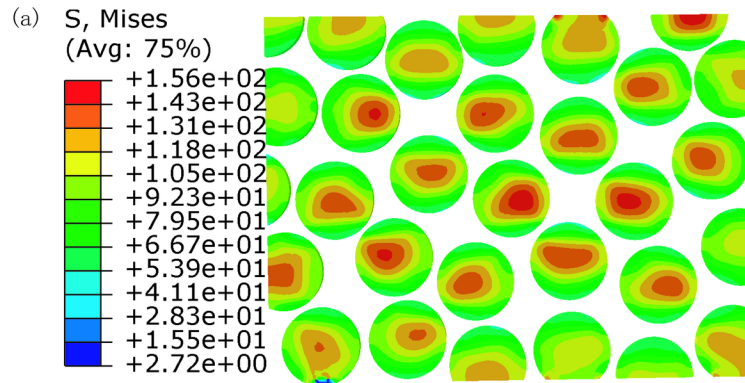


Fig.16 Stress contours of fibers ($\epsilon=0.500\%$) at the $V_f=70\%$ (a) and the $V_f=40\%$ (b) conditions