# Passive ground-based optical techniques for monitoring the on-orbit ICESat-2 altimeter geolocation and footprint diameter

Lori A. Magruder<sup>1</sup>, Kelly Brunt<sup>2</sup>, Thomas Neumann<sup>3</sup>, Bradley Klotz<sup>1</sup>, and Michael Alonzo<sup>4</sup>

<sup>1</sup>University of Texas at Austin <sup>2</sup>University of Maryland, NASA Goddard Space Flight Center <sup>3</sup>NASA Goddard Space Flight Center <sup>4</sup>University of Texas at Austin

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#### Abstract

NASA launched its second Earth observing laser altimeter in 2018 with mission objectives of studying the changes in our climate by monitoring global elevations, particularly in the polar regions. Since the mission is focused on generating accurate elevations and elevation change, the geolocation (or geodetic position) of the measurements are of upmost importance to each of the scientific disciplines supported by these observations. Geolocation validation is required to ensure that the mission is meeting its objectives with the appropriate level of geolocation accuracy. One validation technique uses small optical reflectors placed in a specific pattern along one or more satellite ground-tracks. The optics provide a unique signal back to the satellite that can be used to compare the geolocation of these returns in the data to the known position on the surface. Results of the position comparison indicate the measurement locations are accurate to within 3.5 m with a standard deviation of 1.6 m. They also provide a method for determining a representative footprint diameter using geometric analysis, which resulted in an average value of 10.9 m  $\pm 2.1$  m.

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- <sup>6</sup> <sup>1</sup>Applied Research Laboratories, University of Texas at Austin, Austin, TX.
- <sup>7</sup> <sup>2</sup> NASA Goddard Space Flight Center, Greenbelt, MD
- <sup>8</sup> <sup>3</sup> University of Maryland College Park, College Park, MD
- 9
- 10 Corresponding author: Lori A. Magruder<sup>†</sup> (<u>magruder@arlut.utexas.edu</u>)
- <sup>†</sup> Address: 10000 Burnet Rd., Austin, TX, 78758.

#### 12 Key Points:

- Corner cube retro-reflectors are passive optical components that provide a distinct and
   recognizable reflection signature to space-based lidar for validation of the measurement
   geolocation.
- The passive optics also provide a methodology for determining the diameter of the laser
   footprint on the surface and the effects of the atmospheric attenuation on the effective
   spot size.
- Validation studies confirm that the accuracy ICESat-2 geolocation at both mid-latitude
   and polar regions currently meet the mission requirement for position quality.
- 21

#### 22 Plain Language Summary

NASA launched its second Earth observing laser altimeter in 2018 with mission objectives 23 of studying the changes in our climate by monitoring global elevations, particularly in the polar 24 25 regions. Since the mission is focused on generating accurate elevations and elevation change, the geolocation (or geodetic position) of the measurements are of upmost importance to each of the 26 scientific disciplines supported by these observations. Geolocation validation is required to ensure 27 that the mission is meeting its objectives with the appropriate level of geolocation accuracy. One 28 29 validation technique uses small optical reflectors placed in a specific pattern along one or more satellite ground-tracks. The optics provide a unique signal back to the satellite that can be used to 30 31 compare the geolocation of these returns in the data to the known position on the surface. Results of the position comparison indicate the measurement locations are accurate to within 3.5 m with a 32 33 standard deviation of 1.6 m. They also provide a method for determining a representative footprint diameter using geometric analysis, which resulted in an average value of  $10.9 \text{ m} \pm 2.1 \text{ m}$ . 34 35

#### 36 Abstract

Corner cube retro-reflectors (CCRs) are passive optical components that were used to 37 independently evaluate the ICESat-2 laser altimeter geolocation and laser footprint diameter. 38 These campaigns were performed in both mid-latitude and polar regions over the first 18 months 39 40 of the mission. A proven technique using CCRs to evaluate the original ICES at mission was optimized for the ICESat-2 mission and deployed at White Sands Missile Range and along the 88° 41 S line of latitude to passively monitor the geolocation accuracy, and estimate the diameter of the 42 laser footprint. The results reveal an average geolocation accuracy of the ICESat-2 measurements 43 to within 3.5 m  $\pm 2.1$  m, meeting the mission requirement of 6.5 m. Additionally, the CCR 44 evaluation of the footprint diameter resulted in 10.9 m  $\pm$  1.3 m, with the variability explained 45 primarily through the influence of atmospheric conditions. 46

#### 47 **1. Introduction**

The Ice, Cloud and land Elevation Satellite-2 (ICESat-2) is a NASA Earth observing 48 satellite, on-orbit since September 2018. The motivation behind the mission is focused on 49 observations over the polar regions to support ice-sheet elevation change and sea ice 50 characterization studies. The observations are realized through ATLAS (Advanced Topographic 51 Laser Altimeter System), a lidar instrument designed to provide precise ranging measurements of 52 individual, 532 nm laser photon reflections from the surface of the Earth. These ranging 53 54 measurements, combined with the satellite observatory position and laser pointing determination create a capability to quantify centimeter-scale elevation change over the ice sheets and sea ice 55 freeboard (Neumann et al., 2019). ATLAS uses a single laser to provide 6 altimeter beams at 10 56 kHz. The beams are configured into 3 beam pairs with pairs spaced by ~3.3 km across track 57 (Markus et al., 2017). Within each pair the spots are separated in the along-track direction by 2.5 58 km and 90 m in the across-track direction. Additionally, the pairs provide two distinct energy 59 levels with the 'strong' spot at 4 times the level of the 'weak'. The locations of the weak/strong 60 spots in the 6 beam pattern is dependent on the orientation of the observatory relative to the 61 direction of motion. 62

63 The mission requirements on satellite position and pointing determination ensure that the 64 observations accurately support the mission science objectives with respect to geolocation.

Precision orbit determination (POD) is required to be within 5 cm radial accuracy while the 65 precision pointing determination (PPD) requires laser pointing knowledge to within 3.7 µrad of 66 precision, resulting in a total measurement geolocation knowledge of 6.5 m, or ~2.7 arc sec. The 67 PPD solution quality is dependent on the ability to resolve the influence of thermal variations and 68 spacecraft orientation on the pointing efficacy (Luthcke et al., 2019). Pointing corrections on-orbit 69 are determined using a regular sequence of maneuvers over the ocean ('ocean scans') and within 70 a full orbit ('around the world scans') to recover the range residuals and biases (Luthcke, et al., 71 2005). These calibrations are applied to the PPD to reach the pointing accuracy to support the 72 mission geolocation requirement. Methods for validating the geolocation utilize comparison to 73 ground reference surfaces, derived independently with high resolution airborne lidar (Magruder et 74 al., 2020) or ground based GPS surveys (Brunt et al., 2019) to determine the vertical and horizontal 75 accuracy of the ICESat-2 geodetic position. 76

77 A unique, independent method for geolocation accuracy assessment relies on small optical components, corner cube retroreflectors (CCRs). CCRs are designed to reflect light along the angle 78 79 of incidence with diffraction properties dependent on the CCR diameter and energy wavelength 80 (Sun et al., 2019). As such, careful selection of diameter allows the ground-based optics to return 81 observable reflections to a space-based receiver (Magruder et al., 2020). These CCR signatures are distinct from the surface returns at the measurement rate, meaning no post-processing or 82 aggregation is necessary to identify the presence of reflections from the optics within the data. 83 Comparing the satellite measurement geolocation to the known geodetic position of the CCR gives 84 85 the independent assessment of measurement positional accuracy. This technique was successful for ICESat (2003-2009; Magruder et al., 2005) and the airborne engineering testbed for ICESat-2, 86 MABEL (Multi Altimeter Beam Experimental Lidar; Magruder and Brunt, 2018). Studies with 87 MABEL determined the expected response of a CCR to photon-counting lidar and helped design 88 the specific implementation details. The arrays placed at White Sands Missile Range (WSMR) 89 utilized 8 mm diameter optics on poles with height variations from 0.6 m to 3 m in four arrays 90 (Magruder et al., 2020a). CCR arrays placed along the 88° S (88S) line of latitude during a GNSS 91 ground survey performed in 2017-2018 Antarctic summer season and revisited each year since use 92 CCRs positioned where the satellite reference tracks (RGTs) spatially converge. The WSMR 93 94 arrays are a diamond pattern with CCR arrays located at each of the vertices (north, south, east, 95 and west). This pattern is capable of capturing a beam pair for both ascending and descending tracks. For example, an ascending track could illuminate the east and north arrays with the right beam of a pair and the south and west array with the left beam of the pair. The design of the full ensemble at each CCR validation assumes the ICESat-2 satellite meets the pointing control requirement of  $\pm 45$  m and the ability to maintain RGT tracking (Magruder et al., 2020b).

Here we present an evaluation of the ICESat-2 positional (horizontal geolocation) accuracy 100 of the laser footprint and the estimation of the effective footprint diameter using all available CCR 101 signature returns for the current mission lifetime (October 2018 – May 2020). Further, this study 102 103 explores the variations in the results to radiometric effects due to the state of the atmosphere. The methods presented here provide an opportunity to passively monitor the performance of the 104 105 satellite on-orbit. This method is critical to the continuous validation of ICES at-2, as it helps ensure that the scientific discoveries leveraging higher-level data products are based on accurate and 106 107 precise low-level data.

#### 108 2. Data and Methods

The CCR analysis focuses solely on the ICESat-2 Global Geolocated Photons (ATL03) data 109 product. ATL03 is the Level 2a data product that provides the photon positional data in addition 110 to many other parameters associated with uncertainties, corrections for tides and atmosphere and 111 signal statistics (Neumann et al., 2019a). The ATL03 data used in this study are available through 112 the National Snow and Ice Data Center (www.nsidc.org) and are part of the third release (r003) of 113 final data products from the ICESat-2 Project Science Office (Neumann et al., 2019b). Since 114 ATL03 includes photons associated with both surface reflections and solar background noise, the 115 ATL03 processing algorithm also includes a filtering method that statistically delineates between 116 the two. This is an important step particularly for the majority of the higher, Level 3a, products 117 that rely on those photons identified with a high surface-signal probability to interpret the surface 118 elevation (Smith et al., 2019, Kwok et al., 2019, Neuenschwander and Pitts, 2019). 119

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The ICESat-2 reference ground-tracks (RGTs) create several opportunities to point ATLAS at the WSMR validation site using the up to 5 degree off-nadir pointing capability of the observatory. These 7 WSMR relevant RGTs equate to opportunities nearly twice a month over the course of the repeat ground-track cycle (91 days) and offer scenarios for both ascending and descending tracks. This variability in satellite direction of motion (northward or southward), combine with possible orientation (forward or backward facing) and solar panel position to provide a comprehensive method for understanding the efficacy of the pointing calibration, the pointing control and overall geolocation knowledge of the satellite-based measurements under on-orbit conditions specific to a mid-latitude orbital position.

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The CCR arrays deployed along the 88S line of latitude during a GPS traverse in the 2017-2018 Antarctic field season (Brunt et al., 2019) and resurveyed or relocated during the following two field seasons to optimize the opportunities for CCR illumination based on the results from the previous year. Given the 92° orbit inclination, the ground tracks spatially converge at the extreme latitudes and provide a dense coverage of the region, allowing for many more CCR potential overpass opportunities than the WSMR location.

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The analysis of the CCR signatures is described in Magruder et al. (2020a) and initiates with 138 finding the elevated, linear signatures distinct from the estimated terrain surface in the relevant 139 ATL03 transects. The second processing step extracts the CCR returns and generates a statistical 140 141 estimate of along-track signal CCR length using the expected Gaussian energy distribution of the laser footprint. The laser footprint diameter is representative of a Gaussian beam diameter, as 142 143 defined as the 2-sigma value of the signal distribution curve representative of 86% of the total beam energy based on pre-launch measurements of a Gaussian energy pattern (Martino et al., 144 145 2019). For a strong beam, the Gaussian beam energy value is 120 µJ while a weak beam provides 30 µJ (Neumann et al., 2019a). The CCR returns are quite pronounced but it is estimated that the 146 147 spatial extent of the footprint size is susceptible to atmosphere attenuation or other optical loses. In this paper, we use the same statistically robust methodology of Magruder et al. (2020a) on the 148 149 much larger set of CCR returns collected since that initial study. The chord length (e.g. the effective distance from leading edge to trailing edge of the initial illuminating footprint and final 150 illuminating footprint respectively) is critical to understanding where the CCR is located relative 151 to the center of the laser footprint diameter. Based on the ATL03 geolocation algorithm, the 152 specific position of a reflector within the footprint dimeter is unresolvable as the geolocation for 153 all of the surface-reflected photons for a given shot are geolocated to the laser footprint centroid 154 (Luthcke et al., 2019). The CCR deterministic position within a laser footprint relies solely on its 155 relative position to the footprint centerline that can be quantified by the length of the signal chord 156

length. This is more of an estimation when only one CCR is illuminated and the scenario lacks the 157 geometric constraint for determining which side (east or west) of the footprint centerline the CCR 158 is located. However, when two or more CCR signatures appear in a single transect, both the ATL03 159 geolocation accuracy and the Gaussian beam diameter can be determined. These estimations are 160 derived using the known geometry of the CCR positions and the goodness of fit of the CCR signal 161 signatures for a given beam diameter through an iterative process. By minimizing the horizontal 162 residuals of the combined comparison over the sequence of CCR illuminations a quantitative 163 assessment is achieved for the geolocation offsets in northing and easting that inform the accuracy 164 of the horizontal position of the measurements. 165

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Determination of geolocation accuracy and footprint diameter is dependent on the signal 167 retrieval. Adequate signal retrieval implies that there is reasonable number of signal photons for 168 accurate analysis given the retrieval technique. The number of signal photons depend on both the 169 incident energy at the surface and returned energy at the receiver. The signal reflection strength is 170 primarily dependent on the surface reflectance. However, atmospheric attenuation also is 171 172 significant. Clouds, aerosols and water vapor in the atmospheric column attenuate the laser light through absorption and scattering. Initial predictions based on data prior to ICESat-2 launch 173 174 suggested that column optical depth, a measure of the observed attenuation related to clouds and aerosols, could attenuate > 50% of the laser energy if the optical depth is above 1.0 (Palm et al. 175 176 2020).

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178 To assess the atmospheric effects on the CCR geolocation validation and diameter retrievals, calibrated attenuated backscatter (CAB) and lower along-track-resolution parameters, including 179 180 relative humidity, temperature, and pressure, are utilized from the ICESat-2 atmospheric product ATL09 (Palm et al., 2019). Specific humidity is determined from these variables following the 181 computation of vapor pressure through the standard equations provided in Rogers and Yau (1996). 182 The atmospheric profiles are separated into estimated boundary layer (surface -1 km), low (1 km 183 -5 km), moderate (5 -10 km), and high altitudes (10 -15 km) to pinpoint specific levels where 184 the moisture content is potentially high or where temperature inversions may trap scattering 185 particles. For the cases at WSMR, the column optical depth and profile data are used to assess their 186 effect on footprint diameter and associated geolocation accuracy. For those cases examined at 88S, 187

other parameters are considered from the atmosphere product, including the blowing snow 188 confidence level which could cause additional attenuation through signal scattering. Moisture 189 190 effects on the signal retrieval are more likely at WSMR than at 88S because temperatures are higher and the air can hold more water vapor. The depth of the troposphere at the mid-latitudes 191 extends above 12 km height as well, especially during the summer months. For comparison, the 192 tropospheric depth at 88S extends only to ~8 km before noticeable stratospheric warming begins. 193 Therefore, the impact of the atmosphere will be different for the two locations based on these 194 differences in atmospheric column characteristics. 195

#### 196 **3. On-orbit assessment 2019-2020**

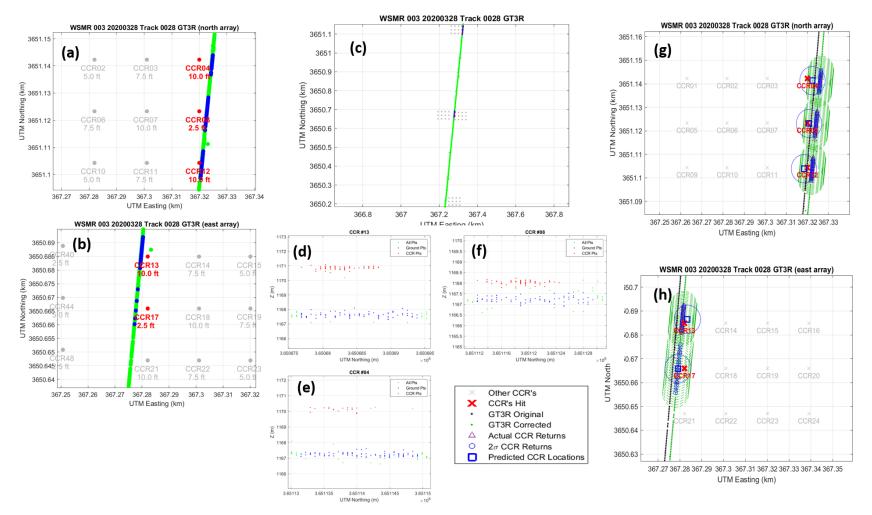
The first successful WSMR overpass in March 2019 captured the center weak beam on 197 both the east and the north arrays. This initial analysis helped assess the geolocation of the ATL03 198 photons relative to the CCR positions for the first time and also helped establish the deterministic 199 method for recovery of the footprint diameter (Magruder et al., 2020a). The result for this particular 200 pass was a footprint diameter of 10.6 m and the ATL03 horizontal geolocation was accurate to 201 within 2 m. This was an important discovery for several reasons that include: 1) the automation of 202 203 the footprint diameter determination which proved to significantly impact the ability to retrieve the correct position of the CCR within the spatial extent of the illuminated areas; and 2) the 204 realization of how the spacecraft pointing is implemented and executed in addition to 205 understanding the constraints on the spacecraft pointing stability. 206

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Favorable conditions in September 2019 allowed for the illumination of multiple CCRs 208 209 with the center strong beam, during the second opportunity to explore the validation technique. The analysis of these signatures indicates a beam diameter of 12 m and horizontal geolocation 210 211 offset of 5 m. The disparity in diameter recovered between the weak (March 2019) and strong (September 2019) beams, at the time, was attributed to atmosphere attenuation and potential loss 212 of return signal at the edges of the footprint where the energy is lower (Magruder et al., 2020). 213 Given the two instances of CCR signal retrievals, the concept and implementation were confirmed 214 but the small sample size of data did not warrant statistically relevant conclusions for geolocation 215 216 accuracy and beam diameter. As such, the pointing requests continued with regularity for the relevant RGTs within each full 91-day orbit cycle. Varied results were achieved at WSMR as some 217 unsuccessful overpasses simply did not hit the CCR arrays while others were performed during 218

cloud cover that prevented returns from either the surface or the CCR. Other opportunities occurred 219 coincident to anomalous satellite operational events which temporarily suspended on-orbit data 220 collection. RGT #28, a descending pass on 28 March 2020 was particularly successful by 221 illuminating 5 CCRs in both the north and east arrays after the satellite performed a roll maneuver 222 to point ~4° off-nadir. Figure 1 (a – c) provides the configuration of the satellite ground track 223 (ATL03 geolocated photons) and the array locations. In Figure 1 the signal photons attributed to 224 the CCR returns are blue and the remaining ATL03 signal photons are green. Figure 1 (d-f) 225 presents the along-track segments for three different CCRs that were illuminated. The heights of 226 the individual signatures provide the initial identification of a specific CCR, as they were deployed 227 with staggered heights with this consideration. In this case of Figure 1 (d-f), the sequential heights 228 of 3 m, 0.75 m and 3 m, from North to South, is indicative of a CCR04, CCR08 and CCR12 229 illumination pattern. The identification of the CCRs in the east array is done similarly (CCR13 and 230 CCR17). The signature chord length of each of the CCRs is statistically derived using a  $2\sigma$  value 231 from a Gaussian distribution of the extracted CCR signal. Using these chord lengths with the 232 geometric constraints from the known CCR array positions results in a 12.0 m beam diameter and 233 a horizontal geolocation offset of 4.3 meters (RMSE 2.1 m). The RMSE is representative of the 234 goodness of fit of the predicted geolocation track to the 5 individual CCR positions. Figure 1 (g-235 h) illustrates the geolocation adjustment of the original track (black line) to accommodate the 236 geometry of the known CCR locations (red X's) and the signal chord lengths (blue points) as an 237 238 indicator of where the CCR is relative to the centerline of the laser footprint. The solution of the true positions (green points) allows the technique to estimate the accuracy of the ATL03 239 geolocation. 240

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- Figure 1. Signal analysis for WSMR overpass on 28 March 2020 where the laser illuminated 5 CCRs within one descending ground track; 3 of the illuminated
- 245 CCRs were in the north array and 2 were in the east. Plots (a-c) show the ATL03 surface signal in green and the ATL03 CCR signal in blue. Plots (d-f) provide the
- returns from CCR04, 08 and 12 in the north array indicating the height of each CCR relative to the surface. Plots (g-h) show the estimated chord lengths (blue) and the solution (green) for determination of the footprint diameter and geolocation offsets that correspond to the geometry of the known locations of the illuminated
- 248 optical components.

WSMR overpass date	Beam Ground- track	Beam strength	Off- nadir angle (deg)	Slant range (m)	ATLAS spot number	Local time of collection	Horizontal geolocation error (m)	Footprint diameter (m)
3/31/2019	GT2R	Weak	2.46	485162	4	08:37	2.5	10.6
5/31/2019	GT2R	Weak	0.37	484743	4	17:04	3.0	8.5
9/28/2019	GT2R	Strong	2.52	485037	3	11:56	5.0	12.0
10/12/2019	GT2R	Strong	4.87	486609	3	23:15	3.8	10.6
10/31/2019	GT2R	Strong	3.27	484847	3	10:24	2.8	11.4
1/07/2020	GT1R	Strong	3.22	485483	5	18:03	0.7	8.3
2/08/2020	GT3R	Strong	1.67	484471	1	16:31	7.7	11.6
3/28/2020	GT3R	Strong	4.06	485998	1	03:16	4.3	12.0
5/09/2020	GT3R	Strong	1.72	484482	1	13:11	2.1	10.0
Average WSMR geolocation error and beam diameter							3.5 ± 1.9	10.6 ± 1.3
88S overpass date	Beam Ground- track	Beam strength	Off- nadir angle (deg)	Slant range (m)	ATLAS spot number	Local time of collection	Horizontal geolocation error (m)	Footprint diameter (m)
12/10/2018	GT2R	Strong	-	511475	3	21:49	0.4	11.6
1/18/2019	GT3L	Strong	-	511930	5	19:17	2.4	9.8
11/27/2019	GT3L	Weak	2.78	512538	2	04:27	3.3	11.1
12/30/2019	GT3R	Strong	2.81	512031	1	02:55	2.0	12.4
1/13/2020	GT2R	Strong	0.67	511625	3	04:47	8.5	11.5
1/24/2020	GT3R	Strong	2.81	512554	1	22:31	3.8	11.5
Average 88S geolocation error and beam diameter							3.4 ± 2.5	11.3 ± 0.8

#### 251 **4. Discussion**

The overpass summary for on-orbit performance of ICESat-2 serves as evidence of the 252 253 mission meeting both the requirement for geolocation accuracy and the pointing control capability. 254 Table 1 provides the results of each relevant case at WSMR and 88S. Aggregation of the results give an average footprint diameter of 10.9 m with a standard deviation of 1.2 m and a median value 255 256 of 11.4 m. The horizontal geolocation error, overall, is 3.5 m  $\pm$  2.1 m. Results specific to WSMR indicate the diameter retrievals are fairly consistent over the 9 cases within a range of 8.3 m to 12 257 m (average value 10.6 m  $\pm$ 1.3 m). Using the 6 overpass cases for 88S independent of WSMR 258 determines an average diameter of 11.3 m  $\pm 0.8$  m. The pre-launch measurements of the ATLAS 259 laser beam divergence were 21.4 µrad and 19.7 µrad for the semi-major and semi-minor axes 260 respectively (A. Martino, pers, comm.). Using these pre-launch measurements and the orbital 261 altitude range of ICESat-2 (486 km - 512 km) yields 9.5 m/10.4 m - 10.0 m/10.9 m for the semi-262 major and semi-minor axes of the laser spots on the surface, in good agreement with our 263 observations. 264

To understand more completely the variability in footprint diameter retrievals we evaluated parameters that might influence the footprint characteristics at the surface. Figure 2 provides a summary of the correlation or lack of correlation between the footprint diameter values and these parameters investigated and is discussed in the subsequent sections.

269 4.1 Satellite Altitude

Since the ICESat-2 orbit is nearly frozen, the satellite altitude has a dependence on latitude. 270 271 Although the altitude at WSMR for each overpass are ~484 km, the slant range associated with the off-nadir pointing angle (Table 1) creates a longer divergence path length. However, the largest 272 off pointing angle theoretically increases the spot size by only 5 cm relative to the nadir. The 273 average altitude at 88S is nearly 30 km higher than at WSMR which would implies that the spot 274 would be larger by ~1 m if solely based on path length. The 0.7 m average diameter difference 275 between the two sites' altitudes is consistent with this expected variation but does not account the 276 potential impact of diameter retrieval relative to along-track alignments with the major or minor 277 axes. The lower variability in diameter values at 88S in comparison to that at WSMR could be 278 279 attributed to the consistency of the collection parameters (altitude and path length) or the lower susceptibility to atmospheric influence based on the reduced vertical height of the boundary layers 280

and the low water vapor content in this region. The relationship between the diameter retrievalsand slant range is provided in Figure 2(d) for all cases.

283 4.2 Atmospheric Parameters

The atmospheric parameters examined are based on the GEOS-FPIT model results used to calculate the wet and dry tropospheric correction to ATL03 photon heights (Palm et al., 2019). Figure 2(a-c) shows column properties comparison for two specific cases at WSMR, while Figure 2(e-f) make similar column-property comparisons between all of the passes at both WSMR and 88S.

289 *Humidity*. The relative humidity (RH) and specific humidity (SH) profiles for WSMR in Figures 2(a-b) indicate large amounts of moisture in the lower atmosphere for the 31 May 2019 290 (8.5 m diameter) case, where SH is > 2 times larger than for the 28 March 2020 (12.0 m diameter) 291 case at 4 km above the surface. Overall, at WSMR, the decrease in diameter with increasing SH 292 confirms that moisture content has a negative effect on detecting surface signal. Footprint 293 diameters for WSMR and 88S are shown with respect to the total SH and RH in Figure 2(f). RH 294 and SH are used here to indicate the amount of water vapor present in the lowest portion of the 295 atmosphere. Figure 2(f) indicates that RH is high throughout the column, but these results are 296 misleading due to the cold temperatures at 88S, which are < -23 °C near the surface for all cases. 297 The average temperature at WSMR near the surface was 13.4 °C. Because of this 36 °C difference, 298 the 88S specific humidity is an order of magnitude lower than at WSMR. Overall, SH for the cases 299 at 88S are nearly the same across each instance regardless of apparent footprint diameter 300 determined with the CCR method and thus do not indicate moisture is having an effect on surface 301 signal at this location. 302

303 Optical Depth. The column optical depth comparison for the May 2019/March 2020 results at WSMR in Figure 2(c) indicates a correlation between increasing optical depth and smaller 304 305 diameters due to increasing attenuation. Initially, the column optical depth was analyzed to 306 determine the presence of cloud and aerosol layers as the most probable parameter for indicating signal attenuation. However, the ATL09 CAB and related parameters capture only limited 307 information on the state of the atmosphere at a given location. Clouds above the 12 km threshold 308 309 are typically not detected and increased moisture in the lower atmosphere is also not well detected. With increased moisture content, the possibility of additional aerosols not observed by the satellite 310

increases as well. Looking over all WSMR cases in Table I, Figure 2(e) indicates that the primary 311 range of optical depths is between 0.05 - 1.05, although there are several cases that exceed 1.05. 312 It is fitting that the cases with the largest diameters have the lowest optical depths, but there is not 313 a clear statistical relationship determined. For the 88S CCRs, the column optical depth is high for 314 the lowest diameter case but remains low to moderate for the other cases. There is potential for 315 optical depth to be useful at 88S due to a lower tropospheric height (i.e., below the ICESat-2 316 threshold) and lack of water vapor in the atmosphere, but there is no apparent trend within this 317 small sample. 318

319 Atmospheric Parameter Summary. The trends described, although based on an overall small amount of data, suggest that moisture content is inversely related to the footprint diameter 320 for mid-latitude locations, where the moisture content values are more significant than in the polar 321 regions. Increased moisture could indicate that there is also an increase in forward scattering that 322 widens the full laser footprint at the surface but lowers the detected energy, relative to clear 323 atmosphere. That is, the scattering creates a higher probability that the signal photons on the outer 324 325 edges of a widened footprint are mistaken for noise or are further scattered. While the connection with atmospheric moisture content at 88S is not apparent, another potential cause of low-level 326 attenuation at 88S is blowing snow in the near-surface layer. Blowing snow acts similarly to an 327 aerosol layer, where the small ice crystals provide both forward scattering and backscattering on 328 329 the laser energy. Although not shown, the case with the lowest diameter had a moderate blowing 330 snow identified in ATL09, but several of the other 'normal' diameter cases also had at least moderate blowing snow detected. Additional data collection in the coming Austral summer might 331 help with a more in depth understanding of the relationship between ICESat-2 footprint diameters 332 and both blowing snow and optical depth. 333

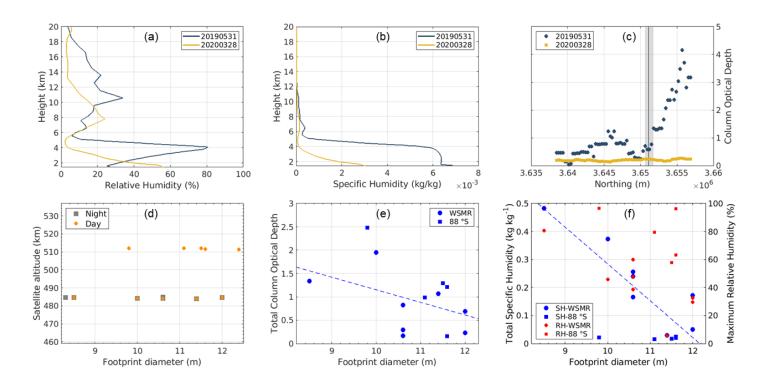
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#### 4.3 Seasonality of Returns from 88S

We note that we only received returns from the Antarctic CCRs during the austral summer months (late November through late January) despite nominal pointing control throughout the calendar year. While it is possible that individual attempts at pointing to a CCR array could be compromised by blowing snow or a snowfall event that attenuates the laser returns, this seems like an unlikely explanation for the total lack of CCR returns over the Feb – November months. We note that the surface reflections during this time period were nominal (Brunt et al., 2019)and suggest that the lack of CCR returns in the Austral winter is due to frost build up on either the optical glass or the plastic cap that holds the optical glass due to near-surface atmospheric supersaturation. This could be tested by relative humidity measurements at the CCR sites, or mitigated by heating the CCRs during satellite overpasses, but we acknowledge the difficulty of making humidity measurements at such low ambient temperatures. CCR signatures are recovered

347 shortly after the sun rises at this latitude, suggesting a thermodynamic explanation.



#### 348

349 Figure 2. Comparisons of relative humidity (RH) profiles, specific humidity (SH) profiles, and column optical depth are provided in (a), (b), and (c), respectively,

for the 31 May 2019 (8.5 m diameter) and 28 March 2020 (12.0 m diameter) WSMR cases. The line and gray shading in (c) mark the location of the CCRs. In (d), footprint diameter is evaluated with the satellite altitude for WSMR and 88S. Day and night designation is provided as well. Panel (e) show the column optical

iolight dialeter is evaluated with the sate and entry work and 885. Day and right designation is provided as well, Failer (e) show the continuous potential

depth and panel (f) shows the total SH and maximum RH in the lower atmosphere with respect to footprint diameter for WSMR (blue circles) and 88S (blue

353 squares). The dashed line in (f) is the best-fit for WSMR cases only.

#### **6. Summary**

We have shown that the passive method of CCR signature analysis provides an assessing 355 and monitoring capability for geolocation validation of space-based laser altimetry. The results 356 from two validation sites indicate that ICESat-2 geolocated photons are accurate to within the 357 mission requirement of 6.5 m. The methodology also provides an assessment of the effect laser 358 spot diameter. To date, the successful satellite overpasses of the sites conclude that the diameter 359 of the ATLAS footprint is  $10.9 \text{ m} \pm 1.3 \text{ m}$ . The variation in independent diameter value retrievals 360 is associated with spacecraft altitude, atmospheric attenuation that can be correlated to optical 361 depth and humidity, or moisture levels. 362

#### 363 Acknowledgments and Data Availability

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#### 371 **References**

- Brunt, K. M., Neumann, T. A. and Smith, B. E. (2019), Assessment of ICESat-2 ice-sheet surface
  heights, based on comparisons over the interior of the Antarctic ice sheet. Geophysical
  Research Letters, 46(22), 13,072-13,078, doi:10.1029/2019GL084886.
- Luthcke, S.B, Pennington, T., Rebold, T., and Thomas, T. (2019), Algorithm Theoretical Basis
   Document (ATBD) for ATL03g ICESat-2 Receive Photon Geolocation, ICESat-2
   Scientific Computing Facility, <u>https://icesat-</u>
   2.gsfc.nasa.gov/sites/default/files/page\_files/ICESat2\_ATL03g\_ATBD\_r002.pdf
- Magruder, L. A., and K. M. Brunt (2018), Performance analysis of airborne photon-counting lidar
   data in preparation for the ICESat-2 mission, IEEE Trans. Geosci. and Rem. Sensing, 56,
   2911-2918, doi: 10.1109/TGRS.2017.2786659.
- Magruder, L., K. Brunt and M. Alonzo (2020a), Horizontal geolocation validation for ICESat-2
   with corner-cube retroreflectors, Remote Sensing, in revision.
- Magruder, L., T. Neumann and N. Kurtz (2020b), Overview of the ICESat-2 mission, Geophysical
   Research Letters, in review.

- Markus, T., and Coauthors (2017), The Ice, Cloud, and land Elevation Satellite-2 (ICESat-2):
   Science requirements, concept, and implementation, Rem. Sens. of Env., 190, 260-273,
   doi: 10.1016/j.rse.2016.12.029.
- Martino, A.J., Neumann, T.A., Kurtz, N.T. and McLennan, D. (2019) ICESat-2 mission overview
   and early performance, Proc. SPIE 11151, Sensors, Systems and Next-Generation
   Satellites XXIII, 111510C; https://doi.org/10.1117/12534938.
- Neuenschwander, A., and K. Pitts (2019), The ATL08 land and vegetation product for the ICESat 2 mission, Rem. Sens. of Env., 221, 247-259, doi: 10.1016/j.rse.2018.11.005.
- Neumann, T.A., and Co-Authors (2019a), The Ice, Cloud and Land Elevation Satellite-2 mission:
   A global geolocated photon product derived from the Advanced Topographic Laser
   Altimeter System, Remote Sens. Of Env., 233, doi: 10.106/j.rse.2019.111325.
- Neumann, T. A., A. Brenner, D. Hancock, J. Robbins, J. Saba, K. Harbeck, A. Gibbons, J. Lee, S.
  B. Luthcke, T. Rebold, et al. (2019b). *ATLAS/ICESat-2 L2A Global Geolocated Photon Data, Version 3.* Boulder, Colorado USA. NASA National Snow and Ice Data Center
  Distributed Active Archive Center. doi: <u>https://doi.org/10.5067/ATLAS/ATL03.003</u>.
  [May 2020]
- Palm, S., Yang, Y., & Herzfeld, U. (2020). Algorithm Theoretical Basis Document (ATBD) for
   the atmosphere, Part I: Level 2 and 3 data products. ICESat-2 Data Products, <u>https://icesat-</u>
   <u>2.gsfc.nasa.gov/science/data-products</u>.
- Rogers, R. R., and M. K. Yau (1996), A Short Course in Cloud Physics. Butterworth-Heinemann
   publishers, pp. 304.
- Smith, B. (2018), Algorithm Theoretical Basis Document (ATBD) for land ice along-track height
   product (ATL06), ICESat-2 Scientific Computing Facility, https://icesat-2 scf.gsfc.nasa.gov/atbd\_docs.
- Schutz, B. E., H. J. Zwally, C. A. Shuman, D. Hancock, and J. P. DiMarzio (2005), Overview of
  the ICESat mission, Geophys. Res. Lett., 32, L21S01, doi: 10.1029/2005GL024009.
- Sun, X., Smith, D.E., Hoffman, E.D., Wake, S.W., Cremons, D.R., Mazarico, E., Lauenstein, J.,
  Zuber, M.T., and Aaron, E.C. (2019) Small and lightweight laser retro-reflector arrays for
  lunar landers, Applied Optics, Vol. 58, No. 33, doi: 10.1364/AO.58.009259
- 415 416